Internal Report No.\_\_\_3\_\_\_

#### Gondola Temperature Study

#### Mark Radomski

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Following is a summary of conclusions reached by studying temperature vs time curves of p  $\alpha$  e and electron gondola balloon flights; and by calculations. Details of data and calculations are available in a notebook labeled "Temperature Study." (References such as 4:2 refer to section 4 in notebook, page 2).

Chart of Numerical Information  $p \alpha e$ 

Source

Неа	t	Cap	'v
11Cu	ч.	oup	J.

Item

 $C = 7 \text{ k cal/}^{O}C$   $C = 4 \text{ k cal/}^{O}C$  4:1 calculation from weight 4:1 lab heat test calc.

## Heater Effect

$$\Delta \dot{T}_{htr} = 5^{\circ}C/hr \text{ at } 30 \text{ W}$$

$$= 7^{\circ}C/hr \text{ at } 30 \text{ W}$$

$$= 4^{\circ}C/hr \text{ at } 30 \text{ W}$$

$$= 4^{\circ}C/hr \text{ at } 30 \text{ W}$$

$$= 4^{\circ}C/hr \text{ at } 60 \text{ W} (!?)$$

$$= 8 \text{ to } 14^{\circ}C/hr \text{ at } 60 \text{ W}$$

$$= 2:2 67C4P \text{ slope change on curve}$$

### Solar Effect

$$\Delta T_{so1} = 3 \text{ to } 10^{\circ} \text{C/hr}$$

$$= 4 \text{ to } 6^{\circ} \text{C/hr} \text{ (AZ)}$$

$$= 1 \text{ to } 2^{\circ} \text{C/hr} \text{ (AZ)}$$

$$= 9^{\circ} \text{C/hr}$$

$$2:1 66C4P \text{ to } 66C5P \text{ difference in cooling}$$

$$2:2 67P3P \text{ to } 67P2P \text{ difference in cooling}$$

$$2:3 67C3P \text{ slope change of curve}$$

$$10:1 \text{ calc. from net heat input of sun}$$

based on 50 W & C = 7

Item

Source

Radiation Loss

 $\left| \Delta \tilde{T}_{rad} \right| = 5 \text{ to } 6^{\circ} \text{C/hr}$  $= 1 - 2^{\circ}C/hr$ 

Convection Loss\*

 $\left| \Delta \dot{T}_{conv} \right| \sim 5^{\circ} C/hr$  $= 4^{\circ}C/hr$  (AZ)  $= 3^{\circ}C/hr$ 

Loss thru Ethafoam

 $U < 1.5 W/^{\circ}C \times (T_{in} - T_{out})$ 

Loss thru Ears

U∠ 3/4 W

Loss during Ascent w/out Sun

$$|\tilde{T}_{asc}| = -9^{\circ}C/hr$$
 1:1 67C2P & 66  
= +8-17^{\circ}C 2:1 66C4P & 66

Window Losses during Ascent

u/ <b>(A</b> 8T)	II	1.5	W	m <sup>-2</sup> (	°K <sup>−1</sup>
U/8T	=	2.5	x	10 <sup>-2</sup>	W/ <sup>o</sup> K
ŕ/бт	=	4.3	x	10 <sup>-3</sup>	$hr^{-1}$

Window Losses due to Rad'n

U/ (AST)	=	$0.5 \text{ W m}^{-2} \text{ o}_{\text{K}}^{-1}$
U/ST	=	8 x 10 <sup>-3</sup> W/ <sup>о</sup> к
т/бт	=	$1.4 \times 10^{-3} \text{ hr}^{-1}$

2:2 67P3P float compared to  $\Delta \tilde{T}_{sol}$ 2:3 67C3P float compared to  $\triangle \vec{T}_{sol}$ 

2:1 66C4P & 66C5P slope changes

2:2 67P3P slope change

2:3 67C3P slope change

6:1 calc. based on 0.95  $W/m^{20}C \times 1.6 m^2$ 

6:1 calc. based on steel screw conductance

C6 P C5 P

7:3 calculation

7:3 calculation

\* Convection was assumed to be "that factor which, present and dominant during ascent, is not present at float altitude."

Chart of Numerical	Information E gondola
Item	Source
Heat Cap'y	
$C = 22 \text{ k cal/}^{\circ}C = 90 \frac{\text{kW - sec}}{\text{oC}}$	5:1 calculation
Heater Effect	
$\Delta \dot{T}_{htr} = 1-2^{\circ} C/hr$ at 60 W	1:3 67P1E, 67P3E
$= 2-3^{\circ}$ C/hr at 60 W	5:1 heat cap'y estimate from wattage
Solar Effect	
$\Delta \tilde{T}_{sol} \sim 3^{\circ} C/hr$	2:4 67C1E compared to 2E, 4E
$\sim 3^{\circ}$ C/hr	10:1 based on 70 W (and heat cap'y)
	from solar effect estimate
Radiation Loss	
$\Delta \dot{T}_{rad} \sim 1^{\circ} C/hr$	p $\alpha$ e average of ~3 <sup>0</sup> /hr scaled by
	$\frac{A_e}{A_p \alpha e} \propto \frac{C_p \alpha e}{C_e} \approx \left(\frac{2.5}{1.6} \times \frac{5}{22}\right)$
Convection Loss*	
$\Delta f_{conv} \sim 1-2^{\circ} C/hr$	scaled from p $\alpha$ e by $\left(\frac{2.5}{1.6} \times \frac{5}{22}\right)$
<u>Loss thru Ethafoam</u>	
$U < 2.4 W/^{\circ}C \times (T_{in} - T_{out})$	6:1 calc based on 2.5 $m^2$ , 0.95 W $m^{-2}$ °C <sup>-1</sup>
Loss thru Ears	
$U < 3/4 W/^{\circ}C \times (T_{in} - T_{out})$	6:1 calc based on steel screw conductance

Loss during Ascent w/out Sun

$$\dot{T}_{asc} = -4^{\circ}C/hr$$
 2:4 67C2E & 67C4E

- Variation of Float Loss/Gain in Sun
  - $\dot{T}_{float} = 0 \pm 3^{\circ} C/hr$  1:4 67C1E

\*Convection loss has been taken to be "that loss which, dominant during ascent, disappears at float."

Item

Source

### Window Loss during Ascent

U/ (AଚT)	=	1.5	₩ m <sup>-2</sup> / <sup>°</sup> K		
U/∂T	=	3.2	$\times 10^{-2} W/^{\circ} K$	7 <b>:</b> 1	calculation
Ť∕ðT	=	1 x	$10^{-4} hr^{-1}$		

Window Loss due to Rad'n

u/(A&T)	$= 0.5 \text{ W m}^{-2}/^{\circ} \text{K}$	7:1 calculation
U/8T	$= 1 \times 10^{-2} W/^{0} K$	
Ϋ́Τ/δΤ	$= 3 \times 10^{-5} hr^{-1}$	

#### Qualitative Conclusions

Consideration of the 66C1P flight which had one inch of ethafoam insulation [3:1] indicates that most losses are thru ethafoam. This conclusion is verified by the calculations of heat loss thru AL ears on base plate [6:1] and thru windows [7:2]. The temperature gradient which exists between top and bottom panduxes of the electron gondola would seem to argue in favor of large losses thru base plate; however, the fact can be explained on the basis of settling of cooler air especially in view of the fact that the gradient was no less in the upside down flight. Thus a third inch of ethafoam might be expected to cut losses by almost 1/3.

Sixty watts of heater necessary and sufficient for  $p \alpha \in [1:2]$ . Furthermore 60 W are not sufficient for E gondola [1:4 and others]. Heater should be pro-rated to surface area - about 90 W for E gondola.

Overheating due to the sun did not occur or threaten in p  $\alpha$  e or E gondola. On the average the sun just compensates radiation losses; it can for limited amounts of time overpower them [see summary "Variation of Float Loss/gain in Sun"] (in 67C2E it overpowered by  $3^{\circ}$ C per hour for 7 hours but this started from  $\sim 0^{\circ}$ C). However, it seems that overheating is unlikely but with a flight like 67C2E it is hard to say what the sun would have done had there been effective heating aboard.

The window is not the major source of loss [7:2]. However, for a gondola with a large window such as  $E^{\pm}$ , it might be worthwhile to improve the window's thermal resistance by

- (1) Aluminum cover over outside of window in ethafoam (painted white on outside) (7:2)
- (2) Intermediate paper barrier to break up dead air space

Making sure that dead space is not thereby made air-tight Pressure dependence of convection may be as weak as  $P^{1/4}$  [9:1]

66C4P - C5P seem to show convection up to 10 MB pressure [2:1]

### Part I

Table of contents, flight summary, temperature conversion, sources for graphed information.

#### Part II

 $p \alpha$  e flight graphs

### Part III

E flight graphs

## Part IV

Misc. results of earlier investigations, but including surface area established for p  $\alpha$  e

### Part V

Sections referenced in conclusions

- 1 Heater effectiveness
- 2 Solar heating
- 3 Insulation
- 4 Heat capacity of  $p \alpha e$
- 5 E gondola heat capacity
- 6 Heat loss mechanisms
- 7 Heat loss from window
- 8 Solar heating calculation
- 9 Pressure dependence of convection
- 10 Heat input effectiveness

# Part VI

Conclusions